# Performance of a Ka-Band Transponder Breadboard for Deep Space Applications

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#### L Introduction

Telecommunication transponders for deep space spacecraft applications [1,2] provide uplink command, turnaround ranging, differential one-way ranging, downlink telemetry, and radiometric capabilities. The Kaband transponder breadboard provides these capabilities at Ka-band uplink [3] frequency of 34.415GI Iz and downlink frequency of 31.977GHz. Operation at Ka-band allows the use of smaller antenna, and provides higher transmission bandwidth to allow 4dB to 6dB improvement over X-band on downlink data rate capability and provides increased accuracy in differential one-way ranging applications. Furthermore, Ka-band has a unique advantage for missions such as Solar Probe, Mercury, and Venus missions as it suffers negligible signal-to noise degradation due to signal scintillation in the solar plasma. The signal-to-noise degradation at X-band is estimated to be about 5 to 10dB in solar flux environment.

To enable flexibility in spacecraft design and modularity in spacecraft hardware, the design was chosen to provide the Ka-trand functionality by adding conversion circuitry to the existing X-band (7.2G11z/8.4G11z) deep space transponder (DS1) design [2] as shown in Figure 1. This modularity allows either X- or Ka-band uplink and X- and/or Ka-band downlink operation.

This article describes the Ka-band transponder design, analysis, and breadboard performance results. The Ka-band transponder block diagram, functional requirements, and design specifications are summarized in Section 11. The transponder phase noise analysis is described in Section 111. The breadboard performance results and conclusions are presented in Sections IV and V, respectively.

### II. Ka-Band Transponder Block Diagram and Requirements

The Ka band transponder functional block diagram and frequency generationscheme are shown in Figure 1. The Ka-band transponder design uses the X-band DST and Ka-band receiver and exciter circuitry to implement the transponder functions at Ka-band uplink and downlink frequencies [2.,3]. The Ka-band receiver and exciter circuits are designed to interface directly with either the X-band DST breadboard or the X-band DST Advanced Engineering Model (DST-AEM) developed by MOTOROLA for JPL. The Ka-band receiver circuitry performs preselection and low noise amplification at Ka-band and downconversion to the X-band DST uplink frequency. The Ka-band excitered circuitry performs telemetry, ranging and differential one-way ranging (DOR) phase modulation on the X-band DST downlink reference and frequency multiplication and conversion to the Ka band downlink frequency.

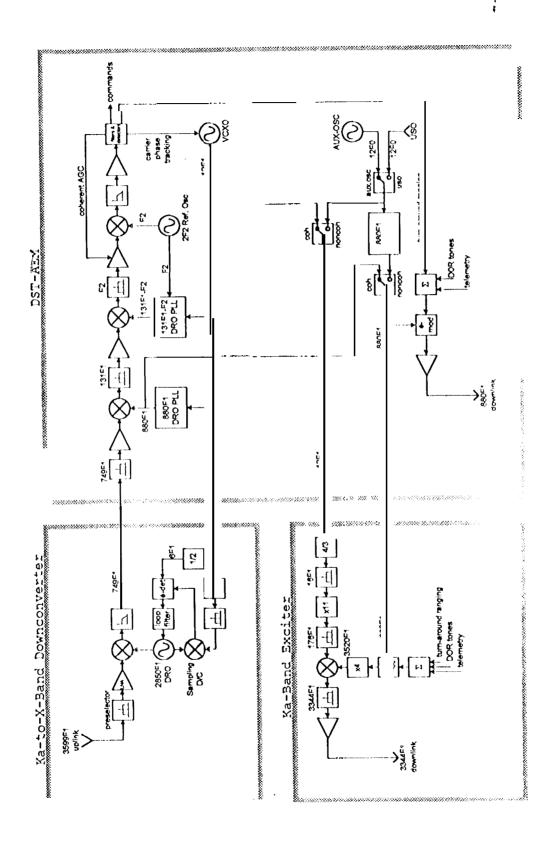


Figure 1. Ka-band transponder functional block diagram

Parameter			
1. Uplink frequency allocation:			
Ka-band uplink	34200 to 34700MHz, deep space		
X-band uplink	7145 to 7190MHz, deepspace		
2 Downlink frequency allocation:			
G-band downlink	31800 to 32300MHz, deep space		
X-band downlink	8400 to 8450MHz, dee, space		
3. Frequency translation ratios:			
Channel 14Ka-banduplink frequency	34415.4375 MHz (3599F <sub>1</sub> )		
Channel 14 X-trend uplink frequency	7162.312S MHz (749 F <sub>1</sub> )		
X-band downlink	S80/3599 (8415 MHz)		
Ka-band downlink	3344/3599 <b>(31977MHz)</b>		
4. Ka-band Receiver Parameters:			
Carrier threshold	≤-149 dBm, unmodulated carrier		
Dynamic range	≥ 79 dB (conserthreshold to -70 dBm)		
Noise figure at Ka-band receiver input	≤6dB		
Acquisition and tracking rate	≥ 550 Hz/sat carrier Level > -110 dBm		
Tracking range	≥ ±100 kHz minimum		
Tracking error	≤ 1°/40 kHz at carrier level> -110 dBm		
Capture range	> ±1 3 kHz at carrierlevel> -110 dBm		
5. Ka-band Exciter Parameters:			
Frequency for coherent operation	3344/3599 times uplink frequency		
Frequency for noncoherent operation	31987.32D988 MHz (3344Fg)  channel16		
RF output power level	≥+3 dBm		
output VSWR	≤ 1.5:1, 50 ± 5 ohms		
Spurious signals	S -60 d Bc		
Modulation bandwidth	≥ 50 MHz @ ± 0.5 dB		
Peak phase modulation index	25 radians at \$ 8% linearity		
Modulation sensitivity	? radians peak/vc, lt peak		
Modulation indexstability	±10% over-20°C to +75°C		
Modulation index:			
- Turnaroundranging	3-9 dB carrier suppression		
- Telemetry	0-15 dB carrier suppression		
-DOR	0-1.1 dB carrier suppression		
Residual phase noise (5 to 25Mt 17)	≤8° rus in the coherent mode		
• , , , ,	S 8° mus in the noncoherent mode		
Carrier phase delay variation	≤ 12 ns over -20°C to +75°C		
Differential phase delay variation \$2 m Ka-/X-band, over -20			
Ranging phase delay variation	$\leq$ 30 ns Over-X3"C to + 75°C		

The design requirements for the Ka-band transponder and frequency translation ratios are. listed in Table 1. The uplink and downlink frequencies are expressed in terms of the reference frequency F<sub>1</sub>. The exact F<sub>1</sub> frequency is determined by the Deep Space Network (DSN) channel in operation. The DSN channel frequencies [2,3] selected for this implementation correspond to channel number 14 at 2F<sub>1</sub> with F<sub>1</sub> equal to 9.5625MHz for coherent mode, and channel number 16 at 2F<sub>0</sub> with F<sub>0</sub> equal to 9.565586MHz for noncoherent mode. The turnaround ratio of the uplink to downlink frequency is equal to 3599/3344 (34.4GHz/32GHz). The criteria [3] used in the selection of the turnaround ratio include maximum possible number of DSN channels for simultaneous X- and Ka-band operation, transmit/receive channel separation for diplexer implementation, central Ka-band allocation considerations, transponder implementation complexity, and uplink/downlink interference considerations

The design requirements shown in Table 1. apply to Ka-bandreceiver and exciter breadboards operating with the X-band DST-AEM. These requirements can be satisfied with no modifications to X-band DST-AEM. The Ka-bandreceiver performance include a noise figure of 5.4dB, an unmodulated carrier tracking threshold of -149dBm, and a tracking range of  $\frac{1}{2}$  100kHz at the assigned channel frequency. The acquisition and tracking rate, is specified to be at least 550Hz/s. The specified nominal output power of the Ka-band exciter is  $\frac{1}{2}$ 3dBm. The exciter output is phase modulated to a maximum phase deviation of  $\frac{1}{2}$ 2.5radians with a radio frequency  $\frac{1}{2}$ 0.5dB modulation bandwidth of 50MHz. The phase noise measured from 51 Iz to 25MHz is required to be less than 8° root mean square (rms) in the coherent mode, and 8° rms in the noncoherent mode.

A. X-band Deep Space Transponder Advancer! Engineering Model Frequency Plan. The functional block diagram for X-band Deep Space Transponder Advanced Engineering Model is shown in Figure 1. The X-band receiver [2] is implemented as a double-conversion superheterodyne phase-lock earner tracking receiver, with a fixed second intermediate frequency (II) The first and second intermediate frequencies are at 131F<sub>1</sub>(1252.7MHz) and F<sub>2</sub>(12.25MHz), respectively. Coherent carrier automatic gain control (AGC) is employed in troth of the IP sections to provide a constant signal plus noise at the carrier loop phase detector. The first local oscillator (I.O) signal at 880 I (8415 MI Iz) and the second LO signal at 1311(1-1;2 are generated hy a dielectric resonator oscillator (DRO) [4] and a surface acoustic wave resonator oscillator (S1{0), respectively. Both of these oscillators are phase locked to the 121 (114.75MHz) voltage controlled crystal oscillator (VCXO). The 121'1 VCXO is in turn phase locked to the uplink carrier. The SROPLL consists of a SRO, a XI 1 multiplier, a ÷ 6 divider, two mixers, a ÷ 2 divider, and a phase detector. The XI 1 multiplier and ÷ 6 divider are used in the SRO 1'1.1. to generate the frequencies 1321'1 and 21'1 from 121'1. The 21'1 and the reference oscillator output at 21<sup>1</sup>/<sub>2</sub> are applied to a mixer followed by a 2 divider to obtain a reference signal at F<sub>1</sub>+F<sub>2</sub> to the SROPLL phase detector. The SRO output at 131F<sub>1</sub>-F<sub>2</sub> and the 132F<sub>1</sub> signals are applied to a mixer to obtain the second input signal at  $F_1+F_2$  to the SROPLL phase detector. A x73 multiplier and a =3 divider are used in the DRO 1'1.1. to generate reference signals at 8361' and 41' 1, respectively.

The coherent downlink carrier at 880F<sub>1</sub> is provided by the LODRO when the DST is operating in the coherent mode from the VCO. In the noncoherent mode, an 880F<sub>0</sub> frequency is generated by the exciter DRO phase locked to the DST 12F<sub>0</sub> auxiliary oscillator (AUX OSC) or the external ultra stable oscillator (USO). The noncoherent downlink signal is automatically selected by the receiver AGC function upon the absence of an uplink signal. An X-band phase modulator is used to phase modulate the downlink signal with telemetry signals, turnaround ranging, or differential one-way ranging tones.

B. Ka-band to X-band Downconverter. The Ka-band receiver circuitry performs preselection and low noise amplification at 3599F<sub>1</sub> (34.415CTI Iz) and downconversion to the 749F<sub>1</sub> @.16GJ Iz) X-band DST uplink frequency. A 5 pole Chebychev waveguide iris preselector filter is used to filter the receiver Ka-band uplink frequency at 34.415G1 Iz, reject the Ka-band downlink frequency at 33441'1 (31.977GHz), and attenuate the receiver LO leakage. After a waveguide-to-coax transition, a MMIC low noise amplifier (LNA) sets the noise figure of the receiver. A triple balanced hybrid mixet is used for downconversion to X-hand.

A phase locked Ka-band DRO generates the 1.0 signal at 2850 F<sub>1</sub>(27.253 GHz) needed for the 3599 F<sub>1</sub> to 749 F<sub>1</sub> downconversion. The phase locked DRO yields lower spurious output (20 to 40 dB) and phase noise than could be achieved with a direct multiplication implementation. An hybrid MMIC sampling downconverter is used to sample the Ka-band LO output with a 36 F<sub>1</sub> reference and downconvert it to 6 F<sub>1</sub> for phase detection with a digital phase/frequency detector. The 6 F<sub>1</sub> and 36 F<sub>1</sub> loop references are derived from the 12 F<sub>1</sub> reference with a divide-hy-two analog Miller divider and a B T multiplier, respectively.

C.Ka-band Exciter. The Ka-band exciter circuitry performs phase modulation on the 880F<sub>1</sub> (8.41 5GHz) X-hand DST downlink reference and frequency multiplication and conversion to the 3344F<sub>1</sub>(31.977GHz) Ka-band downlink frequency. An X-band MMIC phase modulator [5] modulates the telemetry, ranging, and DOR signal on the X-Band DST downlink reference. The modulated X-Band signal is multiplied to 3520F<sub>1</sub> (33.66GHz) with a hybrid x4 varactor multiplier The harmonics are filtered to reduce inband spurious signals on the downlink The 3520F<sub>1</sub> modulated signal is converted to the 3344F<sub>1</sub> downlink frequency by mixing it with a 176F<sub>1</sub>(1683MHz) signal using a MMIC upconverter. The conversion spurious signals are filtered out. The 176F<sub>1</sub> reference for the 'conversion is generated from the 12F<sub>1</sub> reference signal from DST with a Miller divider and step-recovery diode multiplier. The Miller divider multiplies the 12F<sub>1</sub> reference by 4/3 to give 16F<sub>1</sub>. The signal is multiplied with a x11 step recovery diode circuit and filtered to produce the 176F<sub>1</sub> reference.

## III. Ka-Band Transponder Phase Noise Analysis and Results

The downlink phase noise power spectral density of the Ka-band transponder is predicted using noise models [6- I 1] for various devices and the loop transfer fum tions in a custom IPL software package. The phase noise outputs of the local oscillator loops are used in the receiver loop and the exciter to determine the resultant downlink phase noise. The loop transfer functions modify the noise spectral density of the individual contributors when referenced to the loop output. The models for phase noise contributors common to all loops are developed first. These contributors include oscillators, varactors, phase detectors, buffer amplifiers, multipliers, dividers, operational amplifiers, power converter circuits, regulators, and phase modulators. These devices are modelled with the appropriate white phase, flicker phase, white frequency, anti flicker frequency noise characteristics. The phase noise density of each contributor is modified by the transfer function from the contributor to the output. For example, the noise at the phase detector input will be modified by the receiver loop transfer function which is a low-passresponse, whereas the noise at the 12F<sub>1</sub>VCO output will be modified by the loop error transfer function which is a high-pass response. Thereceiver phase noise at the phase locked VCO output is predicted by first calculating the contributions from the Ka-band, X-band, and 1-band local oscillator circuits, and using these phase noise density functions in the receiver loop. The receiver VCO output phase noise density thus obtained is used in the exciter to predict the coherent downlink phase noise density at Ka-band. The Ka-band downlink phase noise densities for the noncoherent AUXOSC mode is obtained by replacing the receiver VCO phase noise density by the AUXOSC oscillator phase noise density. The predicted theoretical phase noise results are compared to the measured results in the following section

#### IV. Experimental Results

The Ka hand transponder breadboard (Figure 1) consisting of the Ka-to-X-band downconverter, Ka-band exciter, and DST-AEM was implemented and performance characterization accomplished. The evaluation measurements include receiver tracking threshold sensitivity, static phase errors for Ka hand uplink frequency offset, swept acquisition characteristics, and AGC versus uplink signal level. All measurements were made at ambient room temperature. The measured tracking threshold sensitivity at the receiver best lock frequency, approximately channel center, is -149dBm. The measured receiver threshold characteristics show good correlation with expected performance (Appendix) overthetracking range as shown in Figure 2. The receiver acquisition characteristics were measured at an input signal level of -110dBm. The measured values for tracking range, and tracking rate are ±1.2MHz at design center frequency, and 800Hz/s, respectively, and meet the specified requirements (Table 1). No receiver false-lock or self-lock resulted during the test phase.

In the coherent carrier mode, residual phase noise is defined for a noise-free received signal case. The phase noise on the downlink carrier signal consists primarily of contributions from the four p}iase-locked oscillators 12H<sub>1</sub>VCO, SRO, X-band DRO, anti Ka-hand 1 DRO used in the Ka-band transponder implementation. Individual phase noise power spectral density functions for these contributors are used in a comprehensive computer program to predict the phase noise of the closed-loop receiver. Total residual phase noise in the output is the sum of the squares all noise sources. The predicted phase noise for the Ka-band transponder in the coherent mode is compared to the measured results in Figure 3. In the intervals between 5Hz and 25MHz on each side of the carrier, the root mean square phase noise is 7° rms, which is below the maximum allowable 8° for coherent downlink. The dominant contributor to this rms phase noise is the 12H<sub>1</sub>VCO and the x4 multiplication process from X-hand to Ka-band, the remaining contributions are less than 10% of the VCO contribution Predicted nns phase noise and Allan deviation are presented in Table 2. The results of the analysis indicate that the coherent mode specifications will be met for both the rms phase noise. The receiver PLL hand limits the VCO spectrum, thus providing the superior performance in the coherent mode.

A comparison of measured to calculated Allan standard deviation characteristics as a function of integration time is shown in Figure 4. The measured Allan standard deviation values for the Ka-band transponder breadboard are in good agreement with the predicted values. Figure 5 shows good agreement between the theoretical and measured phase noise density curves for the noncoherent AUX OSC mode operation.

GaAs MMIC phase modulator [S] developed under SBIR contract at Pacific Monolithics Inc. was used to modulate the X-band reference signal and upconvert (x4 multiplication) it to Ka hand downlink signal as shown in Figure 1. The modulator MMIC chip incorporates a sin~]c-stage input buffer amplifier and a three-stage lumped element hybrid-coupled reflection phase shifters with MESPET varactors to provide a phase deviation of \$12.5 radians with better than 8% linearity. The chip size is 2.49 X 0.91mm Sinusoidal and square

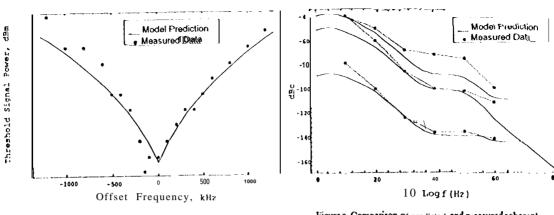


Figure 2. Ka-band transponder carrier tracking threshold vs offset frequency.

Figure 3. Comparison of predicted and measured coherentmode phase noise densities. (Uplink=-100dBm)

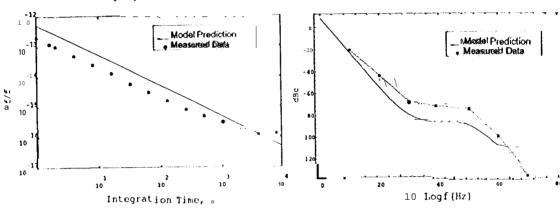


Figure 4. Ka-band transponder Allan deviation vs integration

Figure 5. Comparison of the predicted and measured noncoherent-mode phase noise densities.

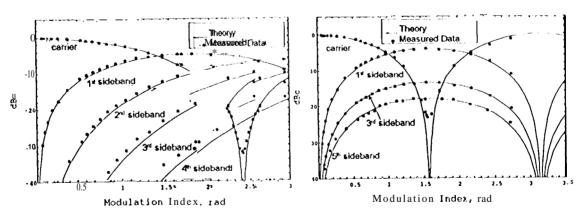


Figure 6. Measured and predicted Ka-band carrier, 1st, 2st, 3st, and 4st sideband levels vs. phase modulation index for the case of a sinusoidal modulating wave of 100-kH 17 frequency.

Figure 7. Measured and predicted Ka-band earner, 1st, 3st, and 5st sideband levels vs. phase modulation index for the case of a square modulating wave of 100-kHz frequency.

"J'able 2 Predicted Phase Jitter and Allan Deviation for Coherent and Noncoherent Modes

			-		
Mode	Frequency	Phase Jitter(5Hz-25MHz)	Allan Deviation		
• • • • • • • • • • • • • • • • • • • •		deg rms	0.01 sec	1 sec	1000 sec
COH ERENT ( P= -1 00dBm)	12F1	0.02	1.510-10	1.3 10-12	9.3 10-16
	880F1	0.80	3.6 10-11	4.6 10 <sup>-13</sup>	4.5 10-16
	3344F1	3.10	- <u>4.6 10<sup>-11</sup></u>	5.3 10 <sup>-13</sup>	4.9 10-16_
NONCOHERENT AUX OSC	12F1	0.08	6.1 10-10	3.2 10 <sup>-11</sup>	3.2 10-11
	880F1	1.98	4.8 10-11	3.? 10 <sup>-11</sup>	3.2 10-11
	3344F1	7.60	5.8 10-11	3.2 10-11	3.2 10-11

modulating waveforms were applied to the phase modulator and their resulting spectra [12,13] were monitored on a calibrated spectrum analyzer. All measurements were performed at ambient room temperature and with a carrier frequency of 31977MHz. A companson of measured and predicted carrier, first, second and third sideband levels for the case of sinusoidal modulation is shown in Figure 6. A modulation frequency of 100kHz was used in the measurements. The peak phase modulation index range was adjusted from 0.2 to 2.4 radians. Figure 6 shows excellent agreement between theory and measurement for sinusoidal phase modulating waves. Negligible amplitude modulation distortion was observed in this case. A comparison of measured and predicted carrier, first, third and fifth sideband levels for the case of square-wave modulation is shown in Figure 7. A modulation frequency of 100kHz was used in the measurements. Good agreement between predicted and measured square-wave modulation results is demonstrated in Figure 7.

#### V. Conclusions

Design concepts and system architecture for a high performance Ka band transponder for deep space spacecraft applications have been presented. The Ka-band transponder has been successfully breadhoarded and evaluated. New technologies such as a Ka-band DRO, X-hand MMIC phase modulator, MMIC amplifiers, MMIC upconverter, and sampling mixers have been integrated into the design. The Telecommunication Development 1 aboratory measurements on the breadboard transponder achieved a threshold level of -149dBm with a dynamic range of 80dB, and excellent acquisition and tracking characteristics. The measured phase noise, Allan standard deviation, and phase jitter data are in good agreement with the predicted characteristics. Measured carrier and relative sideband amplitudes resulting from phase modulation of the Ka-banddownlink signal by sinusoidal and square-wave modulating functions agree well with the predicted results with negligible amplitude modulation distortion.

#### Acknowledgments

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#### References

- [1] J. H. Yuen, Deep Space Telecommunications Systems Lingineering, Plenum Press, New York, 1983.
- [2] N. R. Mysoot, J.D. Perret, and A W. Kermode, "Design Concepts and Performance of NASA X-Band (7162 MI Iz/8415 MHz) Transponder for 1)ecp-Space Spacecraft Applications," TDA Progress Report 42-104, October-December 1990, Jet I'repulsion Laboratory, Pasadena, California, February 1S, 1991, pp. 247-7.56.

- [4] N. R. Mysoor, "An Electronically Tuned, Stable 8415 MHz Dielectric Resonator IET Oscillator for Space Applications," Pmt. IEEE 1990 Aerospace Applications Conference, Vail, Colorado, February 5-9, 1990.
- [5] N. R. Mysoor, and P. Ali, "Miniature X-Band GaAs MMIC Analog and Digital Modulators for Spaceborne Communications Applications," Technology 2001 Conference Proceedings, N A S A Conference Publication 3136, vol. 1, San Jose, California, December 3-5, 1991, pp. 82-88.
- [6] F. M. Gardner, *Phaselock Techniques*, John Wiley and Sons Inc., New York, 1979.
- [7] A. Blanchard, Phase-Locked Loops: Applications to Coherent Receiver Design, John Wiley and Sons Inc., New York, 1976.
- [8] J. P. Frazier, and J. Page, "Phase-I ock Loop Frequency Acquisition Study," IRE Trans. Set-8, September 1962, pp. 210-227.
- [9] D.B.Leeson, "A Simple Model of Feedback Oscillator Noise Spectrum," Pmt. IEEE, Vol. 54, February 1966, pp. 329-330.
- [10] J. A. Barnes, A. R. Chi, and L. S. Cutler, *Characterization of Frequency Stability*, National Bureau of Standards Technical Note 394, Washington, D.C. National Bureau of Standards, October 1970.
- [11] D. W. Allan, "Time and Frequency (Time Domain) Characterization, Estimation, and Prediction of Precision Clocks and oscillators," IEEE Transactions on Ultrasonics, Ferroelectronics, and Frequency Control, Vol. UFFC-34, No. 6, November 1987, pp. 647-654.
- [12] N. R. Mysoor, and R. C). Mueller, "Performance of a 300-Degree Linear Analog Phase Modulator for Communications Applications," Proc. of IEEE in 1993 Aerospace Applications Conference, Steamboat Springs, Colorado, February 1-5, 1993.
- [13] E. A. Whitman, "Phase Modulation Measurement Techniques for Improved Accuracy," Microware Journal, June 1978, pp.113-116.
- [14] F. Stocklin, Relative Sideband Amplitudes vs. Modulation Index for Common Functions using I requency and Phase Modulation, Goddard Space Flight, Center. distributed by National Technical information Service, U.S. Department of Commerce, Springfield, Va, November 1973.

## **Appendix**

The carrier tracking threshold of a 1'1.1. receiver [1,6,7] is defined as the minimum uplink signal required to maintain lock at any given offset from best lock frequency. It is a measure of an important limitation on spacecraft receiver performance.

At the best lock frequency, the carrier tracking threshold signallevel is determined from the following equation:

$$\frac{SI_{i}}{kT'F(2B_{10})} = 1$$

where S is the receiver input signal power level, k is Boltzmann's constant, T is the reference system temperature, F is the receiver noise figure at the transponder input, B<sub>LO</sub> is the one-sided noise-equivalent receiver carrier tracking loop bandwidth at threshold, and 1. is the receiver carrier channelloss. The calculated value of the worst case receiver carrier tracking threshold is equal to -149dBm for a 2B<sub>LO</sub> of 721 lz, channel loss of 1dB, and noise figure of 5.4dB (including isolator, filter, LNA, and connectors) at 290°K.

Note that the carrier tracking threshold signal level can also be expressed in terms of the variance of the phase error due to additive channel noise by

$$\sigma_{\theta}^2 \sim \frac{kTFB_{10}}{sL} = \frac{1}{2}$$

:₹ **.**‡

Therefore at threshold the standard deviation of the phase error is  $\sigma_{\theta} = 1/\sqrt{2}$ . If the uplink signal frequency is offset from the best lock frequency by Af IIz, the phase detector is required to 'operate with a static phase error of  $\theta_{\theta}$  radians givers as

$$\theta_e = \sin^{-1}\left(\frac{2\pi\Delta f}{\alpha K_v}\right)$$

where  $K_{\alpha}$  is the DC gain of the 1'1.1., and  $\alpha$  is the receiver limiter suppression factor given by

$$\alpha = 1 / \sqrt{1 + \frac{4}{n} \frac{kTFB_{if}}{SL}}$$

Here  $B_{if}$  is the noise equivalent predetection bandwidth. This static phase error reduces the threshold phase error standard deviation to  $\sigma_{\theta} = 1/\sqrt{2} - \theta_{e}$ , such that the carrier tracking threshold corresponding to the frequency offset of A f Hz needs to satisfy

$$\sigma_{\theta}^2 = \frac{\text{kTFB}_{LO}}{\text{SL}} = \left(\frac{1}{\sqrt{2}} - \theta_{\bullet}\right)^2$$

The one-side.d noise equivalent carrier tracking loop bandwidth at threshold, B<sub>10</sub>, is given by

$$B_{LO} = \frac{1}{4\tau_2(1-t)/(K\tau_2)} (K\tau_2^2/\tau_1)$$

where  $\tau_1$  and  $\tau_2$  are the loop filter time constants, and K is the open loop gain of the PLL given by

$$K = \alpha K_v \cos(\theta_e) \exp\left(-\frac{1}{2}\left(\frac{1}{\sqrt{2}} - \theta_e\right)^2\right)$$

Note that the open loop gain of the PLL includes factors for the reduction of the phase detector sensitivity due to the static phase error and the additive channel noise.